

AV-20 / AV-20-S

Installation Manual





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Document Revisions

Revision	Date	Description of Change
A	09/20/2017	Initial Release as AeroVonics D-0011-0
В	10/03/2018	Updated for initial NORSEE certification, latest UI for Software
		Release 1.0. Updates from FAA Review.
С	11/24/2019	Release as uAvionix UAV-1003613-001 with updated device part
		numbers
D	11/18/2020	Remove pressurized aircraft limitation. Add guidance for OAT
		installation and AV-20 mounting. Update mechanical drawing.
E	1/21/2021	Update Certification Basis to recognize the clock function.

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1 System Description

The uAvionix AV-20 and AV-20-S Multi-Function Displays provide a wide array of supplemental flight information.

Features Include:

- AoA Display (Voice Alerting & Peaks)
- G-Meter Display (Voice Alerting & Peaks)
- Attitude (Roll / Pitch)
- Slip / Skid
- Clock (GMT / Local)
- Outside Air Temperature (C / F)
- Bus Voltage Display
- Dual User Timers (Count Up / Down)
- Engine Run Timer
- Flight Timer
- Density Altitude Display
- True Airspeed Display (Kts / Mph)
- Internal Battery Operation

The unit incorporates a full color sunlight readable display, bezel-mounted light sensor for automatic display brightness, and an internal battery for operation in the event of power loss. The unit is designed to fit a standard 2¼" mounting hole and is approximately 2 inches deep. Dual ¼" quick-connect fittings are provided for pitot and static connections on the rear of the unit.

A pre-wired wire harness is also available to simplify installation.



The AV-20-S model incorporates internal inertial sensors (gyroscopes and accelerometers), and precision pressure sensors (pitot and static). This allows expanded functionality over the base AV-20 model.

See the detailed functionality matrix to determine which features are available in each model.



Unless explicitly noted, references in this manual to the AV-20 are applicable to both configurations.



2 Model Variants & Required Interfaces

The AV-20 is available in two configurations:

- AV-20 (Part Number UAV-1003591-001 or U-1001-0): The base model <u>does</u> <u>not</u> include internal inertial sensors or pitot-static sensors. Functions are limited to those indicated below.
- AV-20-S (Part Number UAV-1003310-001 or U-1002-0): The enhanced (S)ensor model includes the base model functions plus the inertial sensors and pitot-static sensors. Full functionality is available in this configuration.

Feature / Model	AV-20	AV-20-S	Related Interfaces
Clock	\checkmark	\checkmark	
OAT	\checkmark	\checkmark	OAT Probe ⁽¹⁾
Bus Voltage	\checkmark	\checkmark	
Dual User Timers	\checkmark	\checkmark	
Engine Run Timer	\checkmark	\checkmark	
Flight Timer	×	\checkmark	Pitot and Static Required
AoA	×	\checkmark	Pitot and Static Required
Attitude	×	\checkmark	Pitot and Static Required
Density Alt	×	\checkmark	Pitot and Static Required OAT Probe ⁽¹⁾
True Airspeed	×	\checkmark	Pitot and Static Required OAT Probe ⁽¹⁾
Slip / Skid	×	\checkmark	
G-Meter	×	\checkmark	
Battery Operation	×	\checkmark	
Audio Alerts			
Timer Alert	\checkmark	\checkmark	Audio Panel Connection (2)
AoA Alert	×	\checkmark	Audio Panel Connection (2)
G Limit Alert	×	\checkmark	Audio Panel Connection (2)

Notes:

- (1) The OAT probe is optional. The unit will automatically detect the presence of the sensor and enable functionality related to its use.
- (2) The audio panel connection is optional. Visual alerts will be functional in all modes of operation.

Table 1 - Functional Dependencies





The model and part number are shown on the splash screen on power-up, and on the system setup info page.



Note that the base AV-20 incorporates both pitot and static ports on the rear of unit but are not utilized. This is provided for upgrade purposes and are plugged from the factory. Do not remove the plugs.



Do not connect Pitot and Static in the AV-20 Base model units. Pitot and Static connections ARE required in the AV-20-S Model for proper functionality.

2.1 Equipment Connections

All connections are provided on the single 9-Pin D-sub connector and the two pneumatic fittings.

The unit connects to aircraft power via a normal power circuit with a dedicated 1 Amp breaker.

2.2 Outside Air Temp Input

The outside air temperature interface requires an external analog probe. This port connection is compatible with the Davtron probe analog probe P/N C307PS (not supplied). This is a simple two-wire current source based on the Analog Devices AD590KH component.

The sensor reading may be trimmed in the setup pages.



Do not tap into an existing OAT system that is in use (the probes may not be put in parallel or series).



OAT functionality requires separate approval for installation of the OAT probe.

2.3 Audio Output

The audio panel connection is a low-voltage analog output that is designed to connect directly to an audio panel (typically a non-switched input). High power outputs capable of directly driving a cockpit speaker are not provided.



2.4 Manufacturing Port

A dedicated RS-232 bi-directional serial port is provided for manufacturing test and calibration purposes. These lines are not connected in the aircraft installation. Two spare RS-232 connections are also not connected.

2.5 Pitot and Static Inputs

Internal pressure sensors measure both pitot and static pressure and are required for air data base functionality, including AOA and Attitude. Industry standard ¼" quick connect fittings are utilized. Reference the wiring diagram for details on how to release the tubing from the fitting.

3 Certification Basis

The AV-20 and AV-20-S are both FAA approved designs, are suitable for certificated aircraft and are FAA approved under NORSEE policy PS-AIR-21.8-1602.

The AV-20-S complies with ASTM F3011 Standard Specification for Performance of Angle of Attack System.

When installed in accordance with AC 20-94A, the AV-20 and AV-20-S meet the requirements for an installed clock as specified in 14 CFR § 91.205(d)(6).



The AV-20 and AV-20-S are supplemental and may not be used as a substitution for a certificated aircraft system. Except where otherwise recognized, no operational credit may be taken for installation of this equipment.



The AV-20 and AV-20-S are limited to installation in Part 23, Class I and II aircraft. Installation in other categories of aircraft may require additional certification activities.



Installations requiring penetration of a pressurized aircraft's pressure vessel may require additional certification activities or FAA approved data.



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4 Operating Limits

Operating Limits				
Angle of Attack Range	0° to +30°			
Angle of Attack Resolution	1°			
Angle of Attack Operation	+35 to +300 Knots			
Angle of Attack Accuracy	2.5°			
Density Alt Range (Accuracy)	-1,000 to +25,000 Feet (± 500ft)			
TAS Range (Accuracy)	+35 to +300 Knots (± 20 kts)			
Attitude Angle	No Limits			
Attitude Rate Limit	±250 Degrees / Second			
Attitude Accuracy	1° Static, 2.5° Dynamic			
G Alert Limits	±8g			
OAT Range	-40°C to +70°C			
OAT Accuracy	±4°C			
Slip Range (Accuracy)	±7° (±2°)			
Bus Voltage Range	7 to 35 Volts			
Bus Voltage Accuracy	±1.0 Volt			
Clock Accuracy	± 1 Second/Day			
Timer Accuracy	± 1 Second/Hour			

The following operational limitations are applicable:

Table 2 - Operating Limits





5 System Specifications

Electrical Attributes				
Input Voltage Nominal	+10 to +32 VDC			
Input Voltage Max	+60 VDC			
Input Power Nominal	3 Watts (0.25Amps @ 12VDC)			
Input Power Max	6 Watts (0.50 Amps @ 12VDC)			
Required Circuit Breaker	1 Amp			
Operation on Battery (AV-20-S)	30 Minutes (Standard 15°C Env)			
Physical Attributes				
Mounting Configuration	2 ¼" Round Instrument Hole			
Dimensions wo/Connector	2.4 x 2.4 x 1.2 Inches			
Weight	0.25 Lbs.			
Electrical Connector	9 Pin Male D-Sub			
Pneumatic Connectors	1/4" OD Quick Connect			
Mounting	(4X) #6-32 Machine Screws			
Case Material	High Impact ABS Plastic			
Environmental				
Operating Temp	-20°C to +55°C			
Storage Temp (48 Hrs)	-30°C to +80°C (Via Analysis)			
Humidity (48 Hrs)	90% RH (Via Analysis)			
Optical Characteristics				
Diagonal Size	1.8"			
Resolution	128 x 160			
Contrast Ratio (Typical)	500			
Brightness (Typical)	1000 cd/m ²			
Viewing Angle Left/Right	60°			
Viewing Angle Up	45°			
Viewing Angle Down	10°			
Backlight Lifetime (Typical)	50,000 Hrs			

Table 3 - System Specifications

6 Intended Function

Reference the "AV-20 Pilot's Guide" UAV-1003614-001 Section 2 for intended functionality.



7 Installation

7.1 Overview

Installation consists of the following steps:

- Remove / relocate any old instrumentation
- Add or locate an appropriate power source / breaker
- Wire power and interfaces as needed
- Mount the unit to the instrument panel with supplied screws
- Apply power and perform setup



Proper mounting orientation is important to ensure performance of the AV-20. Confirm the unit is oriented level in the roll-axis when installed in the panel.

7.2 Mechanical Drawing



Figure 1 - Mechanical Drawing



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7.3 Wiring Diagram



Figure 2 - Wiring Diagram



Figure 3 – Unit Connections – DB-9, Male Connector – Rear View

All connections are provided on the single 9-Pin D-sub connector and the two pneumatic fittings.



Pin	Function	Туре	Comment
1	OAT Supply	Output	OAT Sensor Supply Line
2	Serial Input	Input	UAT Display for Traffic
			(Provisional)
3	AV-XPORT Input	Input	Reserved
4	AV-XPORT Output	Output	Reserved
5	Power	Power	+12 to +28 VDC
6	OAT Sensor	Input	OAT Sensor Input
7	Audio H	Output	Altitude Alerts / Other
8	Audio L	Output	To Audio Panel Ground
9	Ground	Power	To Aircraft Ground

 Table 4 – Connector Pinout

7.4 Setup

Refer to "AV-20 Pilot's Guide" UAV-1003614-001 for detailed setup options and procedures. All options are available to the pilot to configure as desired. A summary of the options are as follows:

- Which pages are enabled
- Which voice alerts are provided
- Audio alert volume
- AoA alerting thresholds
- G-Limit alerting thresholds
- Miscellaneous Settings
 - Background color
 - Temperature units
 - Time format
 - o Speed units
 - Screen Popup behavior
- OAT Trim
- Hard Calibration

8 Calibration & Limits

All calibration and limit settings are available to the pilot. These consist of:

- Upper and Lower AoA alerting Thresholds
- Upper and Lower G-Alerting Thresholds
- OAT Temperature Trim
- Hard Calibration

No installation related calibration is required. See Instructions for Continued Airworthiness below for more information on Hard Calibration.



9 Instructions for Continued Maintenance & Operation

9.1 Limitations

- The AV-20 system may be used for supplemental information but may not replace any equipment required under 14 CFR 91.205.
- The AV-20 system is not a required system and may not be used as a substitution for the certificated aircraft system.
- No operational credit may be taken for installation of the AV-20 system.

9.2 Internal Battery

The AV-20 incorporates a small Li-Po battery that may require replacement on an as-desired basis. It is suggested to return the unit to the manufacturer for battery replacement if any of the conditions are present:

AV-20 Models:

• The clock does not correctly maintain time while the aircraft is not operated.

AV-20-S Models:

- The on-battery operation time drops below 10 minutes when operated at nominal temperatures (10°C to 30°C).
- The clock does not correctly maintain time while the aircraft is not operated.

9.3 Hard Calibration

The AV-20-S utilizes internal inertial sensors for multiple purposes. These sensors can drift out of calibration over time. The setup menu provides for a Hard Calibration procedure that re-calibrates these sensors. Perform the Hard Calibration procedure if any of the following conditions are observed:

AV-20 Models:

• Hard Calibration does not apply.

AV-20-S Modes:

• During power on, the AoA reference bar remains flashing (stabilizing state) for more than 2 minutes.



¹⁴ Odem ST. P.O.B. 7042 Petach Tikva 4917001, ISRAEL | Office: +972-3-924-3352 Fax: +972-3-9243385 | sales@hypertech.co.il | www.hypertech.co.il

• During power on, the ALIGNING mode on the Attitude Indicator page remains for more than 2 minutes.



Note the Hard Calibration must be performed while not in flight, and with as little aircraft motion present. Preferred location is inside a hanger with doors closed. The aircraft does not need to be leveled prior to calibration. The Hard Calibration is not accessible while airspeed is above 40 kts.

